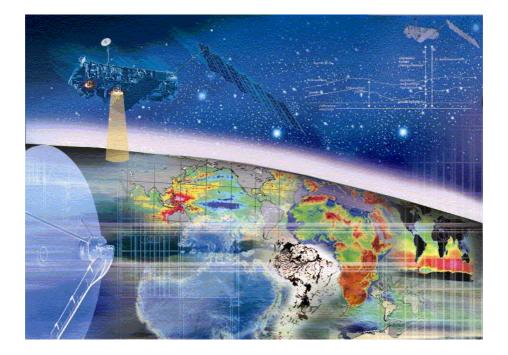




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## ENVISAT CYCLIC ALTIMETRIC REPORT



## CYCLE 76 from 26-01-2009 to 02-03-2009

## **Quality Assessment Report**

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7.2



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## **1 INTRODUCTION**

This document aims at reporting on the performance of the EnviSat Radar Altimeter, Microwave Radiometer and DORIS sensors, on the data quality of the corresponding Fast Delivery products as well as on the main events which occurred during cycle 76.

This report covers the period from 26<sup>th</sup> January 2009 to 2<sup>nd</sup> March 2009.

## **2 DISTRIBUTION LIST**

This report is available in PDF format at the internet address <a href="http://earth.esa.int/pcs/envisat/ra2/reports/pcs\_cyclic/">http://earth.esa.int/pcs/envisat/ra2/reports/pcs\_cyclic/</a>

## 3 ACRONYMS

Auxiliary Data File
Automatic Gain Control
Antenna Pointing Controller
Doppler Orbitography and Radiopositioning Integrated by Satellite
Data Set Record
Electronic Power Converter
European Remote Sensing satellite
European Space Research Institute
European Space Operations Centre
Fast Delivery products
Fast Delivery Geophysical Data Record
Ground Segment
Global Telecommunication System
Height Tracking Loop
Instrument Control Unit
Instrument Engineering Calibration Facility
Intermediate Frequency
Individual Echoes
Instrument Processing Facility
High Speed Multiplexer
Look Up Table
MacroCommand
Main Product Header
Mean Sea Surface
MicroWave Radiometer
Mission Planning System
Microwave Receiver



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NRT	Near Real Time
OBT	On-Board Time
OCM	Orbit Control Mode/Manoeuvres
PCS	ERS Products Control Service
PCF	EnviSat Product Control Facility
PDHS-E	ESRIN Processing and Data Handling Station
PDHS-K	Kiruna Processing and Data Handling Station
PLSOL	Payload Switch-Off Line
PMC	Payload Main Computer
PSO	On-orbit Position
PTR	Point Target Response
RA-2	EnviSat Radar Altimeter bi-frequency
RSL	Resolution Selection Logic
SAD	Static Auxiliary Files
SBT	Satellite Binary Time
SEU	Single Event
SLA	Sea Level Anomalies
SFCM	Stellar Fine Control Mode
SPH	Specific Product header
SPSA	Signal Processing Sub-Assembly
SYSM	Stellar Yaw Steering Mode
S/W	Software
TM	Telemetry
TRP	Transponder
TWT	Traveling Wave Tube
UTC	Coordinated Universal Time
USO	Ultra Stable Oscillator

YSM Yaw Stellar Mode

## 4 **REFERENCE DOCUMENTS**

[R – 1] F-PAC MONTHLY REPORT, SALP-RP-M-OP-XXXX-CN

[R – 2] ENVISAT Microwave Radiometer Assessment Report Cycle, CLS.DOS/07.182, http://earth.esa.int/pcs/envisat/mwr/reports/

[R-3] Envisat RA-2 IF Mask weird behavior: Investigation Report

[R-4] Instrument Performance Evaluation and Analysis Summary, PO-TR-ALS-RA-0042

 $[R-5]\,$  Instrument Corrections Applied on RA-2 Level 1b products, Paper presented at the ENVISAT Calibration Review in September 2002

[R - 6] ENVISAT Phase E Cal/Val Acquisition Plan, ENVI-SPPA-EOPG-TN-03-0008

[R – 7] RA-2 S-Band Anomaly Investigation, PO-TN-ESA-RA-1342, http://earth.esa.int/pcs/envisat/ra2/articles/

[R – 8] RA-2 Performance Results, Paper presented at the ENVISAT Calibration Review in September 2002

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[R – 9a] ECMWF Report on ENVISAT RA- 2 for July 2005, Report on ENVISAT Radar Altimeter - 2 (RA- 2), Wind/ Wave Product with Height Information (RA2\_WWV\_2P), [R – 9b] ECMWF Report on ENVISAT RA- 2 for August 2005, Report on ENVISAT Radar Altimeter - 2 (RA- 2), Wind/ Wave Product with Height Information (RA2\_WWV\_2P), http://earth.esa.int/pcs/envisat/ra2/reports/ecmwf/

[R – 10] Envisat GDR Quality Assessment Report, SALP-RP-P2-EX-21121-CLS015

[R – 11] Envisat RA-2 Range Instrumental correction: USO clock period variations and associated auxiliary file, ENVI-GSEG-EOPG-TN-03-0009

[R – 12] Defining a Rain flag for the Envisat altimeter, G. Quartly, study presented to the final CCVT plenary meeting, <u>http://earth.esa.int/pcs/envisat/ra2/articles/</u>

[R – 13] ENVISAT Weekly Mission Operations Reports # 342-346, ENVI-ESOC-OPS-RP-1011-TOS-OF

 $[R-14] \ Envisat \ validation \ and \ cross \ calibration \ activities \ during \ the \ verification \ phase. \ Synthesis \ Report \ ESTEC \ contract \ No. \ 16243/02/NL/FF \ WP6, \ <u>http://earth.esa.int/pcs/envisat/ra2/articles/</u>$ 

[R – 15] ENVISAT-1 Products Specifications - Vol. 14: RA-2 Products Specifications, PO-RS-MDA-GS-2009, Iss 3, Rev. N, 24/05/2004

[R – 16] Algorithm for Flag identification and waveforms reconstruction of RA-2 data affected by "S-Band anomaly", ENVI-GSEG-TN-04-0004, Issue 1.4

[R-17] Envisat Cyclic Report Cycle 28, ENVI-GSOP-EOPG-03-0011

[R-18] ENVISAT RA-2 IF MASK AUX FILE - Updating Strategy: Investigation Report; C. Bignami and C.Loddo and N. Pierdicca.

## 5 GENERAL QUALITY ASSESSMENT

## 5.1 Cycle Overview

• RA-2 Ku Band is nominal.

RA-2 S Band is no more available. The S Band Power Drop started on cycle 65, on 17<sup>th</sup> January 2008, 23:23:40, UTC (orbit nb 30759). Therefore the S Band parameters, as well as the dual Ionospheric correction in Ku Band are no more relevant and must not be used. Users are advised to use the Ionospheric correction from BENT model, available on FDGDR products;

- The RA-2 Ultra-Stable Oscillator (USO) was nominal on cycle 76;
- Two orbit manoeuvres were executed, on 27th and 28th January 2009;
- The overall number of valid IF masks has been 35, representing 100 % of the acquired and processed IF masks. The auxiliary file RA2\_IFF\_AX has been updated on date 12<sup>th</sup> February 2009;
- During cycle 76, RA2\_SOL\_AX has been updated on the 11<sup>th</sup> February 2009;
- During cycle 76 a Platform anomaly occurred between 15<sup>th</sup> and 16<sup>th</sup> February 2009. This failure caused the on-board software to execute an automatic reconfiguration including immediate transition to Yaw Stellar Mode (YSM). The anomaly can be linked to the appearance of a very bright object across the Star Tracker field of view causing a software



halt. Recovery was successfully performed leading to a transition back to the nominal operating mode. The transition to YSM can be seen in the mispointing plot (Figure 19).

• Tracking performances in the different resolutions are well in line with the output figures and objectives of the Commissioning Phase.

## 5.2 Payload status

#### 5.2.1 ALTIMETER EVENTS

The Radar Altimeter 2, during cycle 76, was never unavailable.

#### 5.2.1.1 RA-2 instrument planning

On cycle 76 IF Calibration has been performed only over the Himalaya site. The operational acquisition has been performed on ascending passes with the NEW procedure for IF Calibration described below. In Figure 1 a map is reported indicating the calibration site.

The II In-Flight Tests aimed to understand the origin of the IF Mask anomaly has been successfully concluded on cycle 66. It has been recognized that the cause of the IF Mask anomaly is the wrong setting of the AGC used for the IF Calibration Mode from time to time. The IF Calibration procedure to be used in operational IF Calibration consists in setting all the AGC's to 3dB before entering the IF Calibration Mode. These parameters are restored to the original values when the IF Calibration mode has expires, before entering in the Measurement mode.

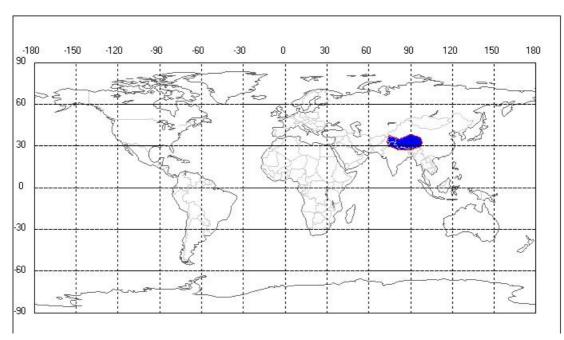


Figure 1: IF Calibration Acquisition sites



The RA-2 instrument planning was performed as follows:

- New procedure for IF calibration (through Digital BITE Mode command) over Himalaya for the entire cycle, 1 ascending pass per day
- No IF calibration on Rocky Mountains.
- Individual Echoes background planning: the buffering of 20 Data Blocks of Individual Echoes (1.114 sec.) transmitted every 160 Data Blocks starts after flying over the Himalayan region (both ascending and descending passes) and is operated for half a day.
- Individual Echoes acquisition (1 second length acquisition, 2 repetitions) over the following sites: Capraia, Toulon D, Vostok, Dome C. Appendix 6 contains a table with the coordinates.
- Individual Echoes acquisitions over the Uyuni Salar
- Preset Loop Output mode for GAVDOS Range transponders, located in Creta.
- Preset Loop Output mode over the ESA transponder located in Rome (permanent location); high chirp resolution.
- Individual Echoes acquisitions during PLO activity over ESA transponder located in Rome (1 second length acquisition, 1 repetition)

Hereafter the map is reported showing the acquisition sites for both the Range and Sigma\_0 transponders.

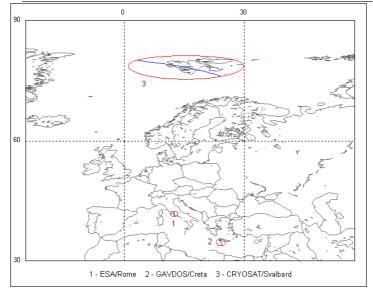


Figure 2A: Transponder Acquisition sites

#### 5.2.2 MWR EVENTS

The MWR, during cycle 76, was never unavailable.



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#### 5.2.3 DORIS EVENTS

DORIS, during cycle 76, was never unavailable.

## 5.3 Availability

The summary of the RA-2 data products availability for the current cycle is reported in Appendix 2. Data availability was around 99.99% for RA-2 products, 99.03% for MWR products and 94.05% for DORIS products.

## 5.4 Orbit quality

During the period covered by cycle 76 two orbit manoeuvres were executed, whose details are given hereafter:

1) On January 27<sup>th</sup>, 2009 (DOY 027) an orbit inclination correction manoeuvre took place. The characteristics of this manoeuvre were:

- Planned delta V size: 1.459 m/s, increasing orbit inclination by approximately 0.01 degree

- Mid thrust time: 04:41:50 utc at ascending node

- Thrust duration: 850.869 seconds

- Measured delta V: 1.459 m/s across track, 0.0012 m/s along track (towards flight direction), - 0.0446 m/s radial (towards downward vertical).

The scope of this inclination correction was threefold:

- keep the ground track within its deadband around the reference one in the high latitudes regions;

- keep the local time of ascending node crossings within the allowed range;

- continue inclination control cycles to allow interferometric applications started with the inclination correction performed on January 23, 2007 and carried over through subsequent manoeuvres on April  $3^{rd}$ , 2007, July 17<sup>th</sup>, 2007, September 27<sup>th</sup>, 2007, December 4<sup>th</sup>, 2007, February 12, 2008, April 22, 2008, July 1<sup>st</sup>, 2008, September 9<sup>th</sup>, 2008 and November 18<sup>th</sup>, 2008.

2) On January 28<sup>th</sup>, 2009 (DOY 028) a one-burn orbit in-plane correction manoeuvre took place. The characteristics of this manoeuvre were:

- Planned delta V size: 0.0085 m/s, increasing the semi major axis by approximately 15 metres

- Mid thrust times: 02:26:00 utc at PSO 343.1 degrees

- Thrust duration: 6 seconds

- Measured delta V: 0.0088 m/s along track (in the flight direction)

The scope of this in plane correction manoeuvre was to keep the s/c ground track within 200 metres from the reference ground track at ascending node over the next few weeks.



## 5.5 Ground Segment Processing Chain Status

#### 5.5.1 IPF PROCESSING CHAIN

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#### 5.5.1.1 Version

Cycle 76 has been processed with IPF processing chain V5.06, installed in both PDHS-E and PDHS-K on  $20^{\text{th}}$  June 2007, orbit 27729.

IPF V5.06 contains the following main evolutions:

- 1. Increase performance in the usage of DORIS Navigator in NRT products due to DORIS Navigator threshold update to 900 seconds coverage RA2/DORIS;
- 2. Alignment of Chain B to Prod Spec 3/N

A complete table of IPF Level1b and Level2 upgrades is reported in Appendix 1.

#### 5.5.1.2 Auxiliary Data File

The Auxiliary files actually used by the IPF ground processing are reported in Appendix 3. The RA2\_POL\_AX, RA2\_SOL\_AX and RA2\_PLA\_AX have been regularly updated without problems. The RA2\_IFF\_AX has been updated during the reporting period. The RA2\_USO\_AX has never been updated during the reporting period. Data are corrected with the RA2\_USO\_AX estimated before the USO Clock anomaly (USO\_Clock\_Period = 12499999726, USO\_Range\_Correction= 17.3 mm).

The RA-2 Auxiliary Data Files (ADF) are accessible from the Envisat Web pages under: http://www.envisat.esa.int/services/auxiliary\_data/ra2mwr/current/

## 6 INSTRUMENT PERFORMANCE

#### 6.1 RA-2 Performance

#### 6.1.1 TRACKING CAPABILITY

The percentages of acquisition in the different resolutions subdivided by surface type are given in the Table 1. The figures given for the RA-2 tracking performances during this cycle are in line with the ones recorded at the end of the Commissioning Phase reported in the second column and presented in [R - 8].



Surface type	320 MHz	Commissioning	80 MHz	20MHz
		Phase objectives		
		320 MHz		
Open Ocean	99.99	>99%	0.01	0.00
Costal Water (ocean depth < 200 m)	98.42	No specific requirement	1.37	0.21
Sea Ice	99.19	>95%	0.70	0.11
Ice Sheet	96.05	>95%	3.28	0.67
Land	80.61	No specific requirement	14.26	5.13
All world	94.98		3.75	1.27

Table 1: RA-2 Tracking capability: Chirp ID percentages discriminated by surface type

In Figure 2, Figure 3 and Figure 4 the cyclic tracking percentages for the three RA-2 bandwidths are reported.

The worsening in performance noticeable for cycle 20 was due to the up-load of wrong on-board software parameters which lasted for about three days whilst for cycle 47 a special operation has been performed to limit RA-2 Chirp Bandwidth to fixed values.

In general, even if a tiny evolution can be observed, the tracking performances are well in line with the output figures and objectives of the Commissioning Phase as given in Table 1.

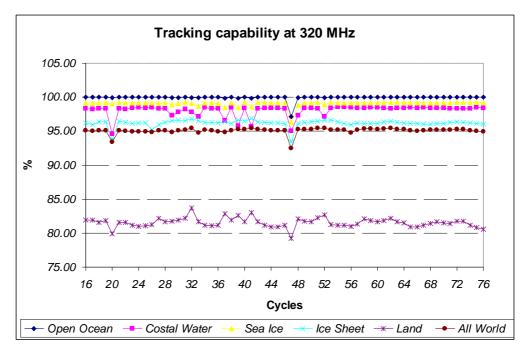


Figure 2: RA-2 Tracking percentage at 320MHz for different surfaces



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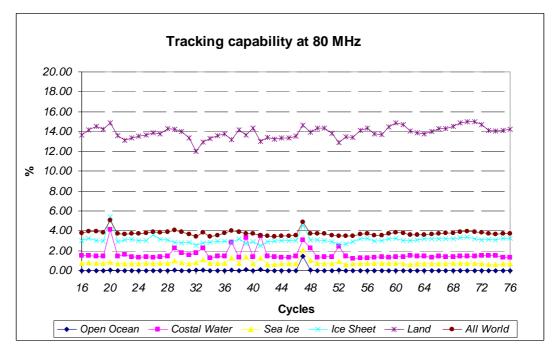


Figure 3: RA-2 Tracking percentage at 80MHz for different surfaces

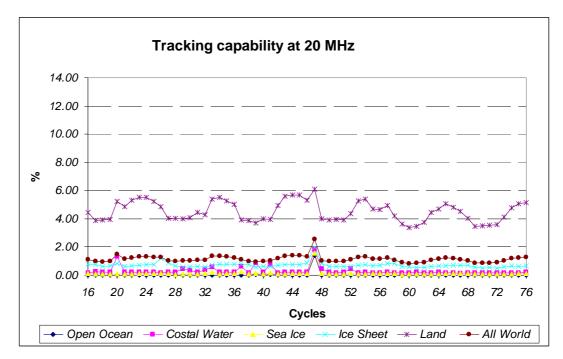


Figure 4: RA-2 Tracking percentage at 20MHz for different surfaces



#### 6.1.2 IF FILTER MASK

In Figure 5 all IF masks retrieved during the current cycle are plotted in the left panel. The onground measured IF mask (ref [R - 4]) is also plotted in that panel with a solid line. In the right panel, the difference of each of the calculated IF masks with respect to the on-ground measured one is reported. The average difference with respect to the on-ground is used as the criteria for defining valid masks: if it is lower than 0.1 db, the mask is considered valid.

According to the planning defined for the IF Calibration acquisition on cycle 76 (ref. Par. 5.2.1.1), one daily pass over the Himalaya (ascending) has been performed with the New procedure for IF calibration.

The NEW procedure consists in setting all the AGC's to 3dB before entering the IF Calibration Mode and resetting all the parameters to the original values before entering in the Measurement mode. It is operationally used since cycle 66 for all IF Calibrations and this ensures 100% of valid IF Masks to be acquired.

The number of IF Masks acquired and processed on cycle 76 was 35. As expected, all 35 IF Masks acquired were valid:

• 100 % of the acquired and processed IF masks were valid.

All the IF masks are used to generate the final IF mask used in the Level 1B ground processing; the method used for generating it consists in a monthly average according to the strategy defined in [R-18] with an editing criteria based on the comparison between each of the single IF masks and the reference one (on-ground).

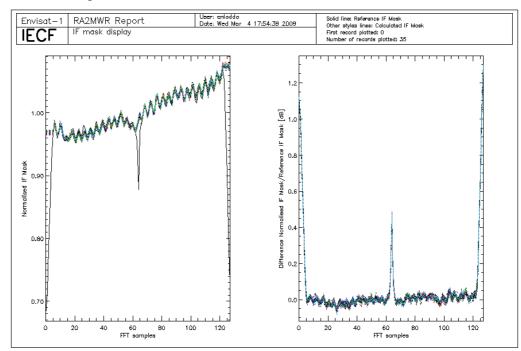


Figure 5: Valid IF masks retrieved during cycle 76 plotted together with the on-ground reference.



In Figure 6 the IF Mask, updated on the 12<sup>th</sup> February 2009, and the previous IF Mask used for processing are plotted.

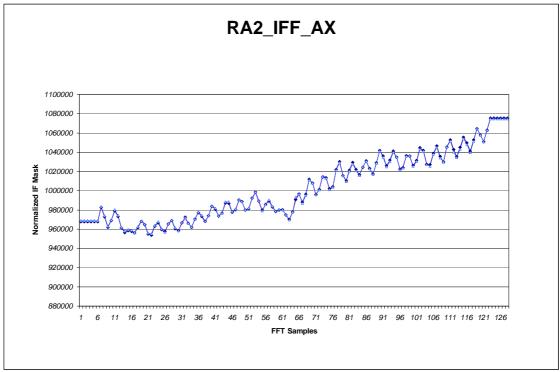


Figure 6: Previous and IF Mask updated on 12<sup>th</sup> February 2009

In Figure 7 the evolution of the IF mask quality parameters evaluated as in [R - 4] is reported only for valid data. It can be observed that the difference with respect to the on-ground reference presents an increasing trend due to the ageing of the instrument. During the current cycle 76 the differences have decreased.

The differences have significantly increased since cycle 56. The masks obtained on the Rocky Mountains present a higher difference with respect to the on-ground mask. This is probably due to the fact that the calibration segments are shorter on this new site and therefore with more noise. However, the difference is always lower then 0.1 db and for this reason the masks are still valid

Some peaks are visible on the plot that correspond to the data acquired on September the 27<sup>th</sup> 2003 at 15:48, on October the 29<sup>th</sup> 2003 at 15:42, on May the 10<sup>th</sup> 2004 at 15:45, on April 9<sup>th</sup> 2006, on December 16<sup>th</sup> 2006 and on September 27<sup>th</sup>. The reason of this could be found in the instrument warming up considering that the IF Cal acquisition has been made, in the three first cases, only a couple of hours after an anomaly recovery. In the two last cases the unavailability was very long, more than two days, and the warming up effect lasted longer. The residual noise and the accuracy show a very constant behavior over the whole period.

During the current cycle the IF Calibration Mode was nominal. The weird behavior described in [R - 3] was no more present. According to the In-Flight Tests performed on cycle 62 63, 64 and 65 this problem, present since the beginning of the mission, seems to be related to the AGC used for the calibration mode.



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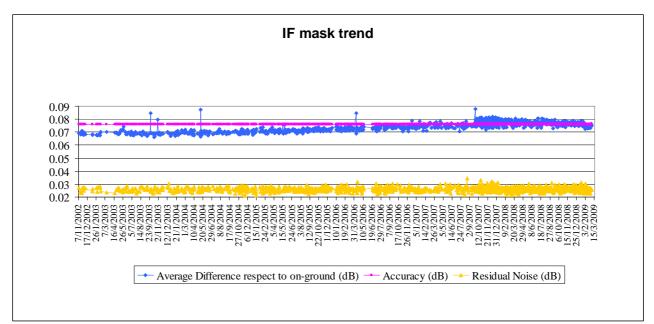


Figure 7: Evolution of the IF mask related parameters for valid IF masks retrieved up to cycle 76

In Figure 8 the percentages of valid IF masks from cycle 20 onwards is reported. This percentage is computed with reference to the acquired masks per cycle. The higher number of valid IF Masks in cycle 48 is a consequence of the special IF Calibration operations which took place on 8 and 9 June 2006 when the altimeter was on its side B. The number of valid IF Masks has decrease from cycle 56 until cycle 61. The high number of valid IF Masks in the last cycles is related to the NEW procedure for IF Calibration Mode applied from cycle 62 onwards, described at the beginning of this chapter. Starting on cycle 66, 100 % of IF Masks were valid because all IF Calibrations were performed using this new procedure.

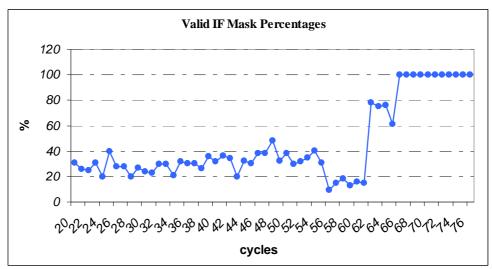


Figure 8: Percentages of valid IF Mask up to cycle 76



#### 6.1.3 USO

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The RA-2 Ultra-Stable Oscillator (USO) was nominal on cycle 76.

In Figure 9 the USO clock period trend is reported. In order to make the variability visible, the difference of the actual USO clock period with respect to the nominal one has been plotted in the upper panel. In the lower panel the Range error due to the USO clock variability has been reported taking a satellite altitude of 800 Km as a nominal value.

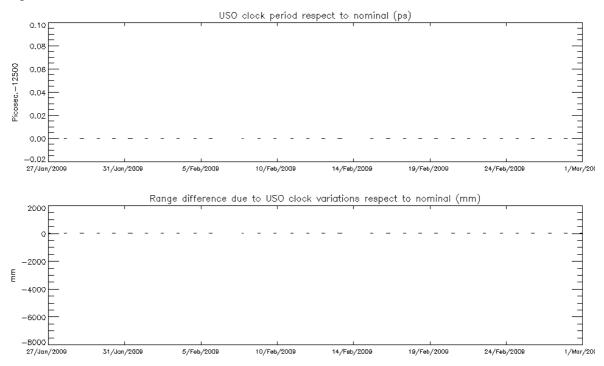


Figure 9: USO clock period (top) and range difference with nominal (bottom) for cycle 76

The USO Clock Period anomaly was almost permanently present during 2006 and 2007. It started in cycle 44, on date 1 Feb 2006 12:04:30, Orbit = 205181. It directly happened after the recovery of the RA-2 on-board anomaly which occurred on the 2006/02/01 at 05:17:56. During the anomalous period, the altimetric range jumped by several meters (about 5.6m) w.r.t. the Mean Sea Surface due to an anomaly in the USO clock period. Moreover, oscillations at the orbital period with an amplitude of 20-30 cm affect the Sea Level Anomaly making the range unusable for both Ku and S Band. The anomaly persisted intermittently until the 15<sup>th</sup> of May 2006 14:21:50, Orbit = 21994, when the instrument was configured to its RFSS B-side. It appeared again when the instrument was reconfigured to its nominal RFSS A-side on date 21 June 2006 13:20:15, Orbit = 22523. The anomaly reappeared after the instrument recovery on date 27<sup>th</sup> of September 2007 11:13:30 and disappeared again for an unknown reason on date 3<sup>rd</sup> of December 2007 03:00:00. The anomaly was back again on the 4<sup>th</sup> of December 2007 13:50:00 and it lasted until the 23<sup>rd</sup> January 2008 14:11:35, orbit nb 30840.



Note that the correction comes back to its nominal value in several steps, causing small uncertainties on the associated correction.

In Figure 10, the USO clock period trend retrieved from the beginning of the mission until the last week of cycle 49 is reported. In Figure 10A, the USO clock period trend retrieved from cycle 50 onwards is reported. The actual value of the USO clock period has been used within the L1b processing; only from the 24<sup>th</sup> of October 2005 (IPF V5.02) until the 1<sup>st</sup> of February 2006. This means that, during this period, the data are corrected for the bias and the drift correlated to the actual USO clock period. The evaluation of the actual USO clock period in this period was performed off-line respect to the IPF processing and it was updated once per week in the auxiliary file RA2\_USO\_AX. The method to correct the data from the USO period changes outside of this period is detailed in Part 7.2.6.

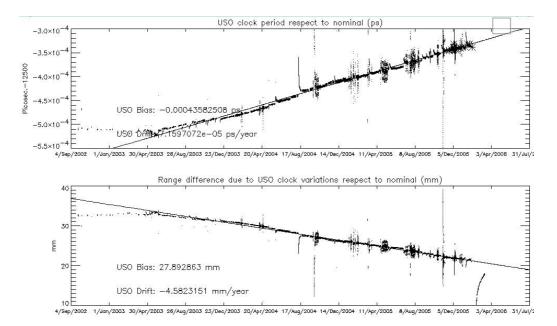


Figure 10: USO clock period (top) and associated range difference (bottom) until cycle 49



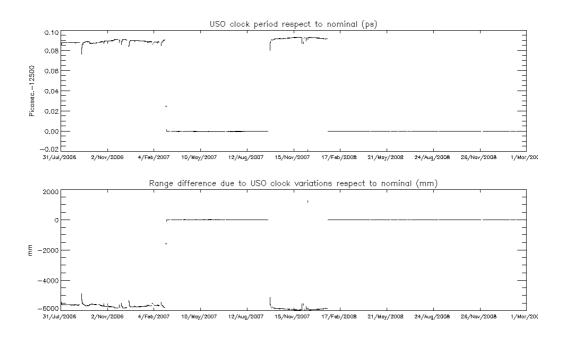


Figure 10A: USO clock period (top) and associated range difference (bottom) from cycle 50 onwards

#### 6.1.4 DATATION

A significant part of an eventual error in the RA-2 products datation could result from imperfect synchronisation between the Satellite Binary Time and the UTC Time due to a drift of the ICU clock period. A correlation between those two times is performed at every Kiruna orbit dump and then extrapolated for the four non-Kiruna orbits. In the upper panel of Figure 11, the differences between the extrapolated UTC values and the corresponding real UTC values measured at the next Kiruna dump, are reported. An anomalous event can be observed in the upper panel plot in between 2008/2009 (mjd 3288) due to the Leap Second correction.

In the lower panel, the ICU clock step for the same period is shown.



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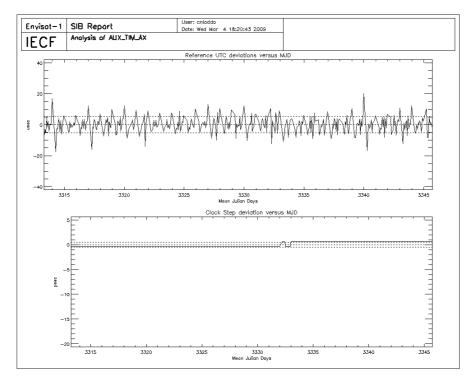


Figure 11: UTC deviations and ICU clock period for cycle 76

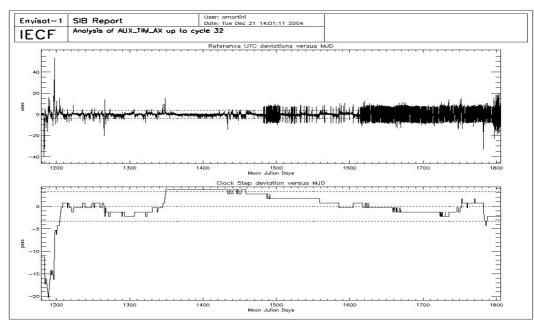
In Figure 12 (upper panel) the differences between the extrapolated UTC values and the corresponding real UTC values measured at the next Kiruna dump, are reported for data up to cycle 32. The UTC deviations for cycle 33 onwards reported in Figure 13.

Only a few anomalous events can be observed at the beginning of the period (cycles 16/17) for which the difference rises above the 20 microseconds warning threshold. However, starting from cycles 22/23, the number of small differences (10 microseconds plus or minus) has increased a lot. Furthermore, during the last ten days of the cycle 32 and for all cycle 33 and 34, the variability of the deviations has increased reporting many peaks just over the 20 microseconds threshold (first part of Figure 12); this phenomenon is now fixed. In the lower panel of both figures the ICU clock step for the same period is shown where big variations are reported. The jump observed around MJD 2288 (07-APR-2006) on Figure 13 is related to the reconfiguration of the Precise Time Correlation process, which became blocked with invalid data after the Service Module anomaly and reconfiguration occurred on 6 April 2006. This is however not a problem because the ICU clock period variations are included in the algorithm for the SBT/UTC correlation evaluation.



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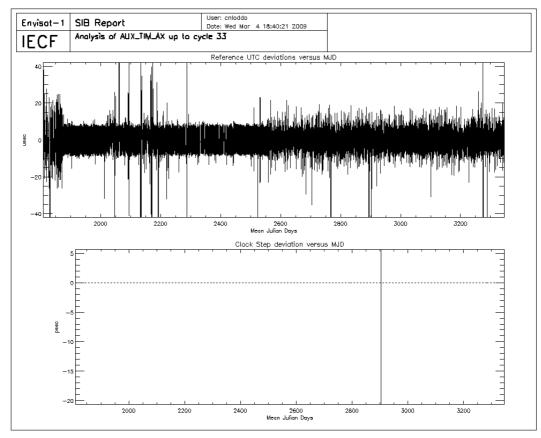


Figure 13: UTC deviations and ICU clock period from cycle 33 onwards



#### 6.1.5 IN-FLIGHT INTERNAL CALIBRATION

The RA-2 Range and Sigma0 measurements are corrected to take into account the internal path delay and attenuation, respectively. This is done by measuring those two variables in relation to the internal Point Target Response. The two correction factors are calculated during the L1b processing and directly applied. They are also continuously monitored and the results for the current cycle (averaged per day) are reported in the next figures. The correction factors on S Band are no more being monitored from cycle 65 onwards due to the lost of the S-band transmission power, occurred on 17 January 2008, 23:23:40 (orbit 30759), see section 7.2.1.

The Ku Band Time delay in-flight calibration factor, reported in Figure 14, shows a regular behavior as observed on previous cycles.

The Ku band Sigma0 calibration factor, reported in red in Figures 15, is nominal.

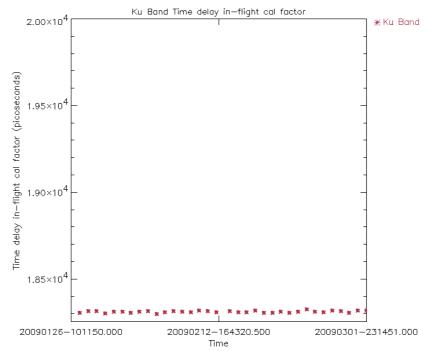


Figure 14: Ku Band in-flight time delay calibration factor for cycle 76 (averaged per day)

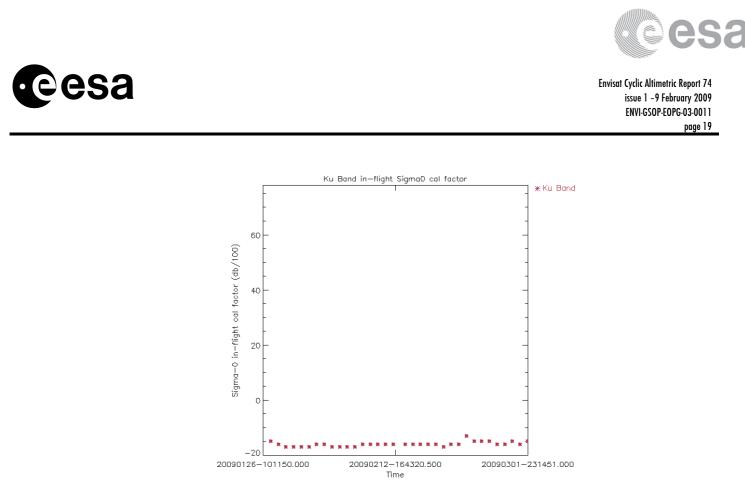
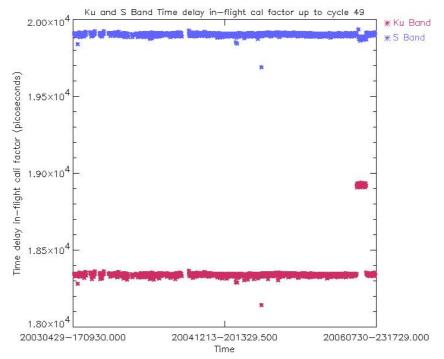


Figure 15: Ku Band in-flight Sigma0 calibration factor for cycle 76 (averaged per day)

Figure 16, Figure 16A, Figure 17 and Figure 17A report Ku and S Band in-flight calibration factors for Time Delay and Sigma0 respectively, daily averaged, up to the current cycle. The Time Delay factor is shown to be very stable for both the working frequencies. The Ku band Sigma0 factor reveals a decrease of about 0.5 dBs over the period starting from cycle 16. As this instability is quite small, it is not being considered a problem for the moment, since the calibration factor is indeed introduced especially to correct for eventual instrumental changes. However, special attention is kept on the monitoring of this parameter. The jump observed on the last part of the plot is related to the period on which the instrument sub-system Radio Frequency Module (RFM) was switched to its B-side, occurred between 15 May and 21 June 2006.

Since Cycle 72 (2 October 2008) a sudden drop of the Time Delay and Sigma 0 in-flight calibration factors has been observed. No impact has been observed on science data. An investigation to understand the origin is on-going.





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Figure 16: Ku and S Band in-flight time delay calibration factor up to cycle 49 (averaged per day)

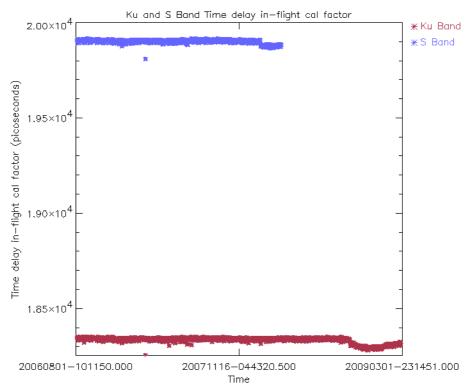


Figure 16A: Ku and S Band in-flight time delay calibration factor from cycle 50 onwards (averaged per day)



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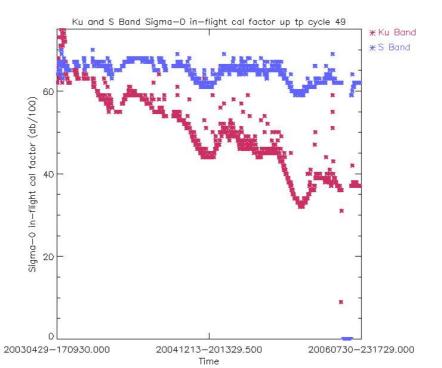


Figure 17: Ku and S Band in-flight Sigma0 calibration factor up to cycle 49 (averaged per day)

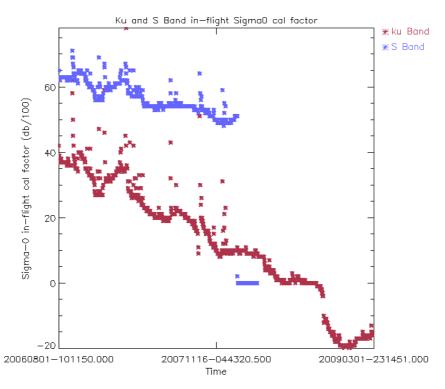


Figure 17A: Ku and S Band in-flight Sigma0 calibration factor from cycle 50 onwards (averaged per day)

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#### 6.1.6 SIGMA0 TRANSPONDER

The  $\sigma^{\circ}$  absolute calibration of the RA-2 is performed using a reference target given by a transponder that has been developed at ESTEC. This has been exploited during the 6 month Commissioning phase to generate early calibration results. In order to consolidate the calibration results and to monitor the RA-2 calibration of  $\sigma^{\circ}$  during the Envisat lifetime, continual monitoring is accomplished by operating the transponder for as many Envisat overpasses as possible. Since the 11<sup>th</sup> of October 2005 the transponder has been moved to a permanent site located in Rome.

The acquisition planned for the 17<sup>th</sup> February 2009 has been successfully performed.

Absolute	Date of	Location / Rel.	RA-2 resolution	Transponder Bias	ECMWF Wet
Orbit nb	Measurement	track		[dB]	Tropo. Corr. [dB]
36441	17th February 2009	Rome/315	High	0.97	0.0803

Appendix 4 reports the transponder measurements from cycle 24 onwards.

The mean value of the estimated bias at High Resolution is 1.00 dB with a standard deviation of 0.09 dB. It is possible to notice that the Low Resolution measurements are coherent among themselves but there is a bias with respect to the High Resolution ones. This is due to a processing problem with the internal calibration factor not taken into account in Low Resolution Mode.

In Figure 18, the time behavior of the bias is plotted for both Low and High Resolution. The green line represents the corrected bias for the internal calibration factor (only for the Low Resolution data) and the tropospheric attenuation. The latter is estimated by using the ECMWF meteorological data. The low value of the corrected bias for the orbit 14397 is due to the dew air condition and a probable underestimation of the tropo-attenuation.



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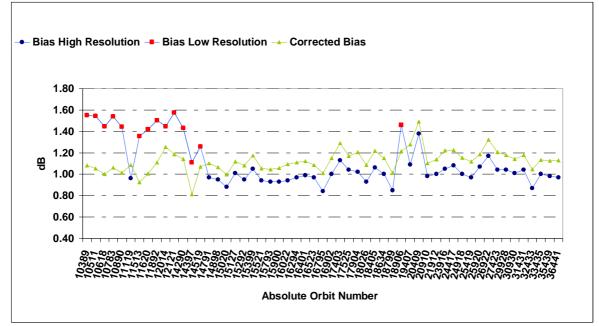


Figure 18: Time behavior of the transponder bias

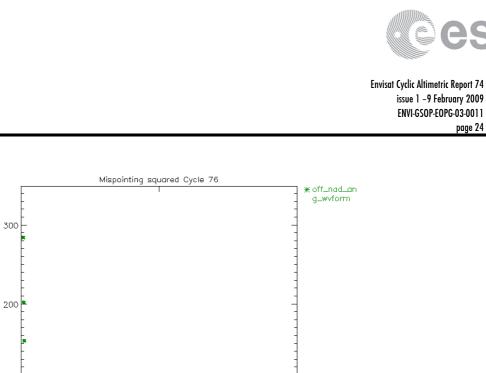
#### 6.1.7 MISPOINTING

In Figure 19, the trend of the mispointing squared (averaged every orbit) is reported in  $deg^{2*10e-4}$ .

The average squared mispointing value, as extracted from the FDGDR data products, has decreased from about 0.028 deg<sup>2</sup>, to 0.0075 deg<sup>2</sup>. This is due to the new algorithm currently used to retrieve the mispointing value from the RA-2 waveform data, see section 5.5.1.1.

Since IPF version 5.02, the mispointing is estimated through the waveform trailing edge slope using an optimum and fixed gate and no longer an adaptive window as defined previously. This allows avoidance of the filter bump effect that leads to high values of the mispointing.

The anomalous values presents at the beginning of Cycle 76 (27<sup>th</sup> and 28<sup>th</sup> January) are due to satellite manouvres, while the bump occurred between the 15<sup>th</sup> (03:38:20 UTC) and the 16<sup>th</sup> (13:09:00 UTC) February is due to a Platform anomaly. This failure caused the on-board software to execute an automatic reconfiguration including immediate transition to Yaw Stellar Mode (YSM). The anomaly can be linked to the appearance of a very bright object across the Star Tracker field of view causing a software halt. The recovery procedure consisted of a switch OFF-ON cycle for the affected Star Tracker followed by star programming verification in YSM. Recovery was successfully performed leading to a transition back to the nominal operating mode.





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Mispointing squared deg2\*1.e-4

1.00

Figure 19: Smoothed mispointing squared trend for cycle 76 (deg^2\*10^-4)

In Figures 20 and 20A, the overall mispointing squared trend (averaged over each orbit) is plotted for cycles 16 onwards.

The low values at the end of the first plot are related to the acquisition in RFFS B-Side, occurred between 15 May and 21 June 2006.

The jump which occurred on date October  $24^{th}$  is related to the upload of IPF version 5.02. The abrupt decreasing of the mispointing squared value is related to the new algorithm, as described in the previous paragraph.

The jump which occurred on November the 26<sup>th</sup> 2003 is correlated to the upload of IPF version 4.56; the abrupt decrease of the mispointing squared value is due to the usage of a new RA2\_IFF\_AX IF mask auxiliary file. After the drop a very tiny increase of the mispointing squared could eventually be detectable. The most probable cause of this phenomenon could be a change in the Intermediate Frequency Filter slope due to ageing effects. For this reason, the RA2\_IFF\_AX will be updated regularly, once per month.



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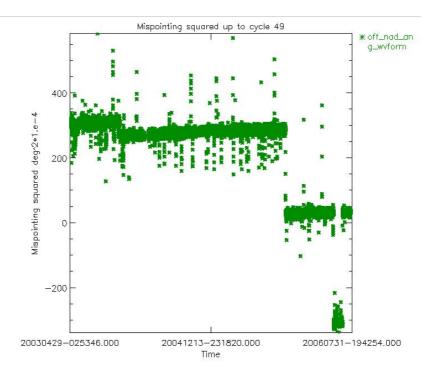


Figure 20: Smoothed mispointing squared trend until end of cycle 49 (deg^2\*10e-4)

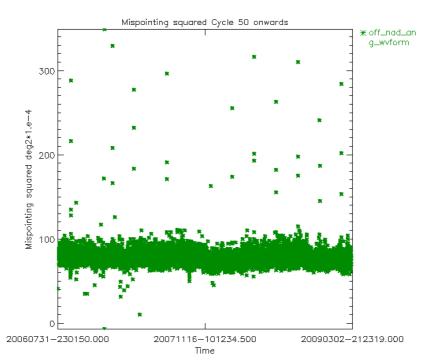


Figure 20A: Smoothed mispointing squared trend until from cycle 50 onwards (deg^2\*10e-4)





It can be noticed that the mispointing squared assumes lower values just after an instrument anomaly, showing an increasing trend until it reaches a standard mispointing value.

This problem has been reduced with the introduction of the updated mispointing retrieval algorithm as described in the previous paragraph.

This particular behavior has always been explained by the different shape that the over-ocean average waveform has before and after an anomalous event as visible in Figure 21, i.e the disappearance of the small dip in the waveforms acquired after the anomaly. Since the new strategy of updating once per month the RA2\_IFF\_AX file, the small bump is not present anymore in the waveforms, see Figure 21\_A, so a new explanation is currently under investigation.

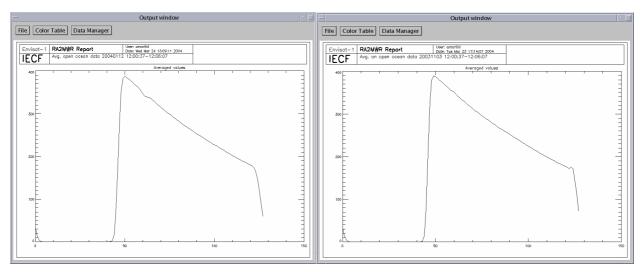


Figure 21: Open Ocean average waveforms before (left) and after an anomaly (right)

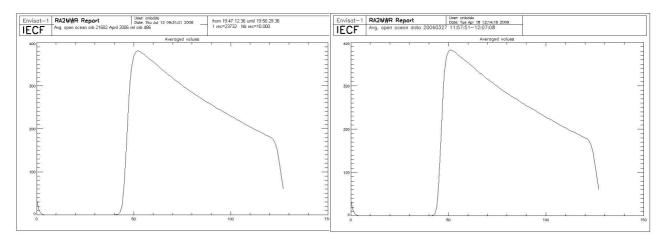


Figure 21\_A: Open Ocean average waveforms before (left) and after an anomaly (right)



#### 6.1.8 S-BAND ANOMALY

Due to the S Band Transmission failure occurred on cycle 65, on 17 January 2008, the S Band parameters are no more relevant. The S Band Anomaly, which consists in the accumulation of the S-Band echo waveforms, are therefore no more present. For this reason, the plot on Figure 22 will no more be updated.

The Patch that prevents the S Band Anomaly by correcting the SW/HW malfunctioning has been successfully uploaded on 27th of June. The Patch has been uploaded for the first time on  $16^{\text{th}}$  of January 2007, but it has been dismissed on  $9^{\text{th}}$  of April because it was causing the Instrument to switch down to Heater 0/Refuse Mode. An investigation has been carried out and some parameter monitoring thresholds causing the instrument to switch down have been modified.

The method used for the identification of the "S-Band anomaly" is statistical and requires a minimum of 1000 seconds of data over ocean. This choice is supported by the fact that the "S-Band anomaly" is associated with a particular instrumental behavior that cannot appear and disappear within a short time frame. (ref. [R - 7])

A valuable algorithm to detect the RA-2 DSRs affected by the S-Band anomaly within the L2 products can be found in [R- 12]. Note that the algorithm is only valid for data acquired over open-ocean.

The IPF version 5.03 includes an algorithm that can detect the presence of the so-called "S-Band anomaly" over any surface. In case of S-Band anomaly detection, bit 1 of the L1b products MCD is set to one; the anomaly is properly detected in 99.9% of the cases.

In Figure 22, the percentage of data per cycle that are affected by the so-called "S-Band" anomaly is reported. The figures are variable between 0% and 8.1%.

The number of occurrences of the S Band anomaly decreased from a mean value of 4% to 2% from cycle 31 onwards due to the implementation of the IF CAL procedure (including Heater 2 for S Band anomaly suppression) twice per day over the Himalayan region.

The relatively high value recorded for cycle 27 is due to the fact that on the day  $1^{st}$  of June 2004, the S-band anomaly started at around 14:30 while the instrument didn't switch to mode Heater 2 when foreseen (at about 15:50). For this reason the S-Band anomaly continued for the next 24 hours until the next Heater 2 mode on June the  $2^{nd}$ .



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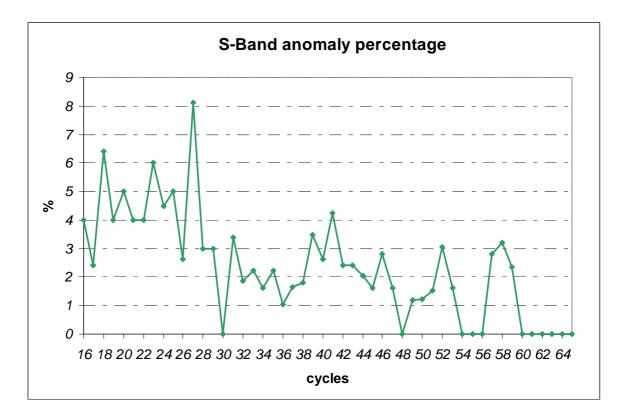


Figure 22: Percentage of data affected by the "S-Band Anomaly" for cycles 16 up to cycle 65

## 6.2 MWR Performance

For MWR performances please refer to the Reference CLS Cyclic Report of the type of [R - 2].

## 6.3 DORIS Performance

For DORIS performances please refer to the Reference F-PAC Monthly Report of the type of [R - 1a] and [R-1b].

## 7 **PRODUCT PERFORMANCE**

#### 7.1 Product disclaimer

A summary of the products released to users and disclaimers on product quality have been established for some products and are available in the following web link: <u>http://envisat.esa.int/dataproducts/availability/</u>





## 7.2 Data handling recommendations

#### 7.2.1 S BAND POWER DROP

Ten hours after the recovery of the HSM anomaly on the 17 January 2008, a drop of the RA2 Sband transmission power occurred. The drop occurred in the South Atlantic Anomaly, showing similar characteristics as for the RA-2 RFSS Side B S-band power drop anomaly occurred in May 2006.

Consequently, all the S-band parameters, as well as the dual ionospheric correction are not relevant and MUST NOT be used from the following date: 17 January 2008, 23:23:40, UTC, orbit nb 30759.

Users are advised to use the Ionospheric correction from Bent model, which is available in FGDR data products:

FGDR (Ionospheric correction from model on Ku-band: field #47)

Investigations have been conducted and the failure of the S Band power stage is considered to be permanent.

#### 7.2.2 SEA-ICE FLAG

The following algorithm is proposed for the determination of a sea-ice flag, presently missing in the Level 2 Ra-2 and MWR data products. (See [R - 14]):

|Latitude (lat: field#4 of L2 data)| >50 deg AND The number of 20Hz valid data (num\_18hz\_ku\_ocean: field#23 of L2 data) < 17 OR |MWR Wet Tropospheric Correction (mwr\_wet\_tropo\_corr: field#42 of L2 data)–ECMWF Wet Tropospheric Correction (mod\_wet\_tropo\_corr: field#42 of L2 data)| > 10 cm OR Peakiness (Ku\_peak: field#139 of L2 data) >2

#### 7.2.3 OCEAN S-BAND ANOMALIES DETECTION

A valuable algorithm to detect the Level 2 DSR affected by the RA-2 S-Band anomaly is proposed in [R-12].

#### 7.2.4 WARNING ON IPF 4.56 VERSION IDENTIFICATION FIELD

All RA-2 and MWR level 1B and NRT Level 2 products generated after November 26, 2003 report a software version as being 4.54 (available in MPH field 8).





Nevertheless those products have been generated with the IPF V4.56 operational since November 26, 2003. The first nominal generated product, using the new SW version, will be the one relevant to the absolute orbit number 9094.

The software version ID is correct since December 4, 2003.

#### 7.2.5 S-BAND BACKSCATTERING COEFFICIENT

For the data processed with IPF version 4.56 on, the S-Band Backscattering coefficient has been demonstrated to be on average about 0.65 dBs higher than for the previous versions of the processor. This is due to the algorithm used for the retrieval of the AGC in S-Band, corrected in IPF version 4.56 to be closer to the real functioning of the instrument.

An average value of 0.65 dBs is suggested to be added to the old software versions S-Band Sigma0 in order to be in line with the new IPF V4.56 version.

#### 7.2.6 USO RANGE CORRECTION

Three different periods can be distinguished:

1<sup>st</sup> period

From the beginning of the mission until the 24<sup>th</sup> of October 2005 the Nominal USO clock period has been used in the processing. The data was not corrected for the bias and the drift correlated to the actual USO clock period. All data acquired before 24<sup>th</sup> October 2005, beginning of cycle 42, have thus to be corrected using the old correction files available on the web site:

http://earth.esa.int/pcs/envisat/ra2/auxdata/OldCorrection.html.

The measured Range shall be corrected considering a drift of -4.58 mm per year and a bias of 29.6 mm.

**Warning for data acquired before cycle 42**: bias and drift have to be **SUBTRACTED** from the original altimetric range, according to the following equation:

#### **Rtrue=Roriginal-dR**

where Roriginal is the range in the GDR products and Rtrue is the true (corrected) range.

#### 2<sup>nd</sup> period

From the 24<sup>th</sup> of October 2005 until the 13<sup>th</sup> of March 2006, outside of the anomaly periods, the actual USO clock period has been used within the processing. The data was corrected for the bias and the drift correlated to the actual USO clock period. Those values, translated into altimetric range figures, are respectively of 28.5 mm and -4.58 mm/year as calculated with data covering the period 13 June 2003 to 01 February 2006.

#### 3<sup>rd</sup> period

From the 13<sup>th</sup> of March 2006 onwards, and during the early occurrences of the USO anomaly, data have not been corrected with the proper value of the USO Clock period. All data acquired during



this period have thus to be corrected using the new correction files. Three USO corrections have been developed for the different Envisat Level 2 altimetry data products for correcting the abnormal RA-2 USO behaviour affecting the Altimetric Range by few meters w.r.t. the Mean Sea Surface:

- A NRT orbit based USO correction for FDGDR products, available from <u>http://earth.esa.int/pcs/envisat/ra2/auxdata/NewCorrection.html;</u> or ftp://ftp.esrin.esa.it/pub/RA2\_MWR/USO/auxdata/
- An Interim daily USO correction for IGDR products, available at the same F-PAC location as for IGDR, in the directory igdr\_ous\_corr
- An OFL cycle USO correction for GDR products, available at the same F-PAC location as for GDR, in the directory gdr\_ous\_corr.

Warning for data acquired after 1<sup>st</sup> February 2006: This correction has to be ADDED to the Ku and S Band altimetric range.

The USO smoothing factor of the NRT USO Correction Tool for periods of NO-USO anomaly has been updated from value 90\*10<sup>-6</sup> ps to value 120\*10<sup>-6</sup> ps starting from 28<sup>th</sup> November 2008 at 12:00

A software routine has been developed to allow users to insert the RA-2 Ultra-Stable Oscillator (USO) corrections into Envisat Level 2 altimetry data products and is available in the same web site than the new correction files.

<u>WARNING</u>: Users are still advised to apply the correction auxiliary files even during the non-anomalous period in order to correct for the nominal ageing drift of the USO device.

#### 7.2.7 KU-BAND BACKSCATTERING COEFFICIENT CALIBRATION

The results of the Ku-Band Sigma0 absolute calibration performed with a transponder have been presented in par. 6.1.4. Those results are going to be consolidated and are summarized in appendix 4, table 18. In order to absolutely calibrate the backscattering coefficient given in the RA2 L2 products, the following shall be used by the end user to get to the real Sigma0 measurement:

#### Sigma\_0\_true = Sigma\_0\_prod + G\_tx\_rx\_prod - G\_tx\_rx\_real - Bias [dB]

Where:

Bias: Bias retrieved from the Sigma0 Absolute Calibration (see 6.1.5)

- **G\_tx\_rx\_prod**: Current effective Tx-Rx Gain value used in the operational ground processing chain (ADF file RA2\_CHD\_AX). The value nominally used since IPF V4.54 is (for configuration RFSS=A and HPA=A) is 170.70 dB
- **G\_tx\_rx\_real**: Pre-launch characterization value (configuration value RFSS=A and HPA=A is 167.46 dB)





#### 7.2.8 RA-2 RADIO FREQUENCY MODULE SWITCHED BACK TO A-SIDE

The Envisat RA-2 sensor has been successfully reconfigured on its nominal side (RFSS A-side) and was commanded back into Measurement Mode on June 21, 2006 at 13.20.15.000 UTC time, Orbit = 22523 after a switch to RFSS B-Side on 15 May at 14:21:50 UTC Orbit=21994 The analysis of the RA-2 data shows an expected behaviour of the RA-2 parameters but also confirmed the persistence of the abnormal RA-2 Ultra-Stable Oscillator (USO) behaviour affecting the Altimetric Range by few meters.

Data from 15 May 2006 until 21 June 2006 was acquired with RFSS B-side and on-ground data processing has been performed with ADFs configured for A-side. For this reason data should be used with maximum care.

## 7.3 Availability of data

#### 7.3.1 RA-2

In Figure 23 and Table 9 (Appendix 2) the summary of unavailable RA2 L0 products is given.

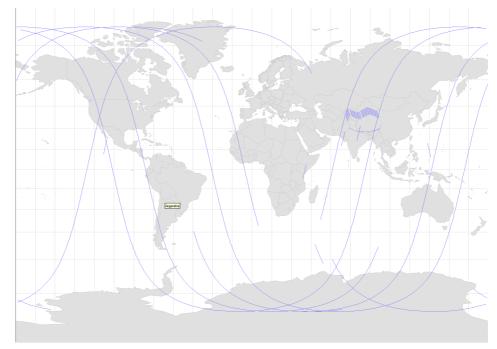


Figure 23: RA-2 L0 unavailable products for cycle 76



It is easy to notice that close to the Himalaya region two small gaps are present per pass, one of about 77/80 seconds and another of about 10 seconds. This is due to the daily instrument switch-offs (Heater 2 mode) performed as part of the IF calibration commands sequence.

Another recurrent gap of around 179-180 seconds is due to an instrument anomaly under investigation. It occurred 11 times on cycle 76.

Apart from these systematic small gaps, several complete orbits are missing on cycle 76 due to ARTEMIS unavailability between 7<sup>th</sup> February at 22:10 UTC and 8<sup>th</sup> February at 6:00 UTC.

In Figure 24 and Table 11 (Appendix 2) the summary of unavailable RA2 L1b products is given. Starting from cycle 65, the reported gaps, both in the map and table, are related to the RA2 L0 products and not to the Mission Planning. No gaps in RA2 L1b data with respect to RA2 L0 products are observed for cycle 76.

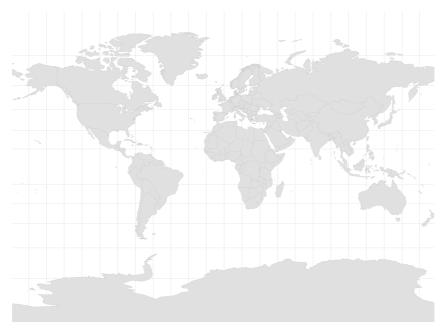


Figure 24: RA-2 L1b unavailable (with respect to RA2 L0) products for cycle 76

Hereafter the percentage of the different levels of products availability is reported.

Considering as reference the instrument unavailability, it is possible to notice that since cycle 66 the situation is improved for all levels of products. The low percentage of data from cycle 56 until cycle 59 is due to the high number of RA-2 Instrument Unavailability occurred as side effect of the SPSA Patch uploaded to prevent the S Band anomaly.

An improvement can be observed on the data coverage since cycle 69.



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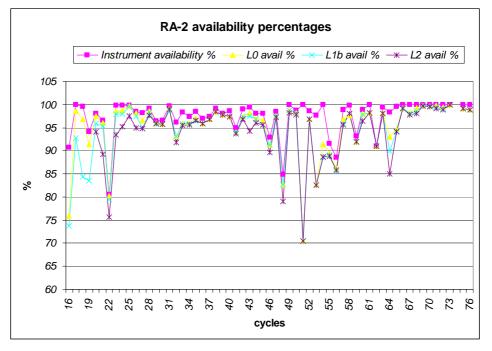


Figure 25: Percentage of Products unavailability

#### 7.3.2 MWR

In Figure 26 and Table 10 (Appendix 2) the summary of unavailable MWR L0 products is given. Several missing orbits can be seen on cycle 76, due to ARTEMIS unavailability between 7<sup>th</sup> February at 22:10 UTC and 8<sup>th</sup> February at 6:00 UTC.

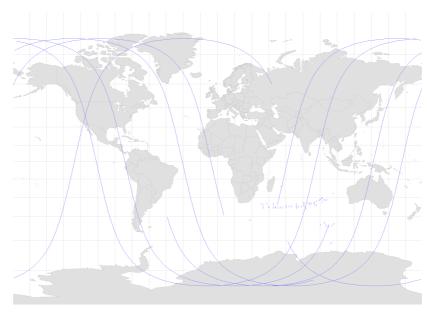


Figure 26: MWR L0 unavailable products for cycle 76



### 7.4 RA-2 Altimeter Parameters

Hereafter a summary of the main Altimetric parameters performances is reported; these results have been obtained using only ocean surface type for the NRT product type, FDGDR.

### 7.4.1 ORBIT

Since the 20<sup>th</sup> of June 2007, operations date of IPF version 5.06, the DORIS Navigator data usage within the NRT processing has increased and it is now being nominally used for 99% of the products. On Table 3 the filenames of the FDGDR products processed without DORIS Navigator on the current cycle are reported. During the reporting cycle all the FDGDR products have been processed with DORIS Navigator.

FDGDR	
none	

#### Table -3: FDGDR products processed without DORIS on cycle 76

The usage of the DORIS Navigator data within the NRT processing increases the quality of the FDGDR SLA, leading to a SLA variability of about 50 cm (to be compared to 20 m when using predicted orbital information).

#### Feature of the ENVISAT Altimetry NRT processing:

The ENVISAT downlink strategy leads to the non-availability of one DORIS Instrument Source Packet (ISP) per orbit for the NRT operations, which corresponds to around 5 minutes per acquired orbit (e.g. number N). The ISP data are not lost but dumped during the following orbit (number N+1, i.e. too late for the NRT ground processing of orbit N).

The ISP DORIS gaps occur in the southern Atlantic (for ascending passes) and in the Indian Ocean and Asia (for the descending passes), Figure 1.



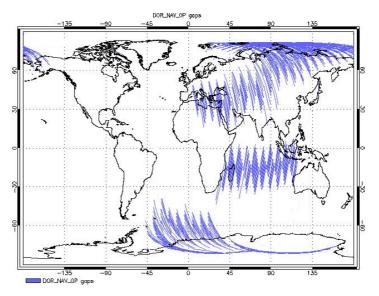


Figure -27: DORIS Navigator missing ISP NRT

In the ground segment processing of NRT data, the last orbit state vector available in the DORIS Navigator is extrapolated until the end of the RA2 product being processed. Given that the extrapolation is not as accurate as the interpolation, the quality of the SLA in the part of the FDGDR products not covered by DORIS Navigator results to be degraded.

An orbit quality flag is available in RA2MWR L1B and L2 products as follows:

• L1B RA2 MDSR: bit 0 of MCD (field 14)

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- L2-RA2 MDSR: bit 27 of MCD (field 8)
- L1B/L2-MWR MDSR: bit 1 of MCD (field 8)

The Orbit quality flag is set to 1 when DORIS Navigator has been used in the MDSR processing.

The Orbit quality flag is set to 0 when the DORIS Navigator has not been used in the MDSR processing and this might happen in two different cases:

- 1. DORIS Navigator is not used on the RA2 processing and the state vector available in the MPH is propagated to the full orbit. In this case all MDSRs in the products present flag set to 0;
- 2. DORIS Navigator is shorter then RA2 due to the dump issue explained above, in this case only MDSRs not covered by DORIS Navigator are set to 0.

The distinction between these two cases can only be performed in a file basis, by checking the SPH field named DS\_NAME="ORBIT\_STATE\_VECTOR\_FILE". The filename is provided for case 2 whilst *NOT USED* is reported for case 1.



### 7.4.2 ALTIMETER RANGE

On Figure 27A, it can be observed that the altimetric range was nominal on cycle 76.

The gaps observed in Figure 27A are related to some PDS failures which prevented the usage of DORIS on NRT products or due to the use of the FOS restituted orbit files (AUX\_FRO\_AX) and to the ARTEMIS unavailability occurred between the 7<sup>th</sup> and the 8<sup>th</sup> February 2009.

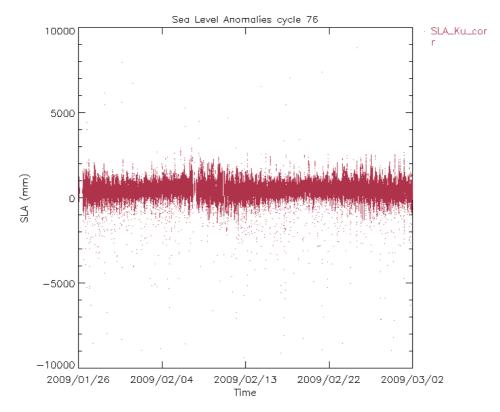


Figure 27A: Sea Level Anomalies Cycle 76 only MDSRs with valid DORIS Flag

The Ku Band SLA has been computed with the following corrections: MWR\_WET\_TROPO, DRY\_TROPO, INV\_BMETER\_HEIGHT, SEA\_KU\_BIAS, IONO\_CORR

Due to the S Band Power drop anomaly, started on the 17<sup>th</sup> of January 2008, see section 7.2.1, the ionospheric correction used for the computation was the Bent model ionospheric correction

In Figure 27B the Histogram of Sea Level Anomalies is reported for the Ku Band. Only MDSRs processed with DORIS have been considered. The peak of the histogram is slightly less than 0.5 meter as expected.



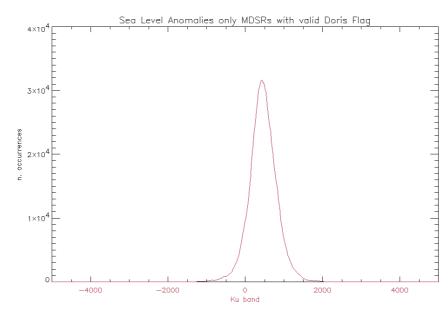


Figure 27B: Histogram of Sea Level Anomalies on Ku Band computed on MDSRs with valid DORIS Flag

### 7.3.3 SIGNIFICANT WAVE HEIGHT

The histogram of the SWH reported in Figure 28 shows a nominal behavior on the Ku Band. The S Band is no more being monitored due to the S Band Power drop, which started on cycle 65,  $17^{\text{th}}$  Janury 2008, see section 7.2.1. The trend goes on following the behavior as detected for the previous cycle. The largest peak (about 60000 data for SWH = 0m) was removed from the plot in order to have the complete picture of the SWH histogram.

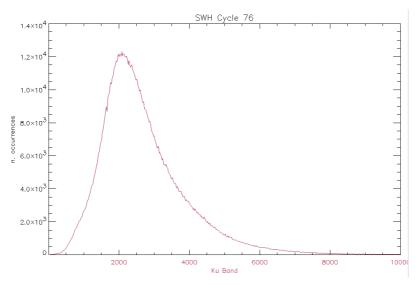
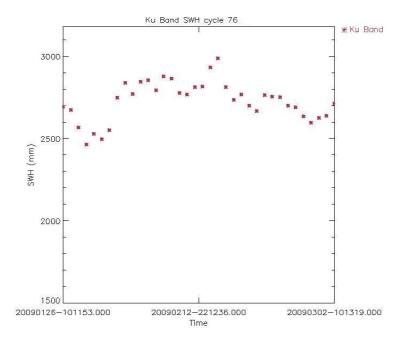


Figure 28: Histogram of Ku Band SWH for cycle 76





esa

Figure 29: Ku Band SWH daily average for Cycle 76(mm)

Figure 29 shows the Ku Band SWH daily mean for the current reporting cycle. The SWH long term plot is reported in two plots, cycle 16 until cycle 49 on Figure 30 and cycle 50 onwards on Figure 30A. It can be noticed that the SWH in both bands shows a trend which follows the seasonal variability. The high daily averages reported (sometimes plotted outside the figure's range) are due to the so-called S-Band anomaly (ref. par.6.1.8).

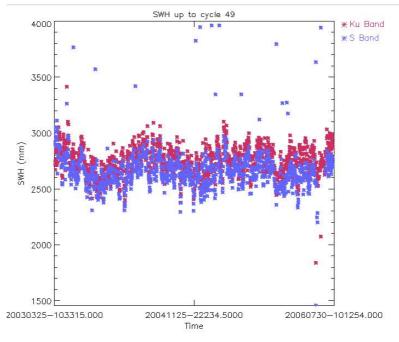


Figure 30: Ku and S SWH daily average up to cycle 49 (mm)



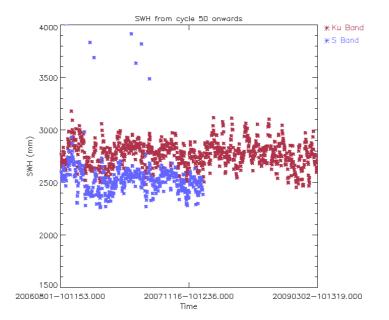


Figure 30A: Ku and S SWH daily average from cycle 50 onwards (mm)

#### 7.3.4 BACKSCATTER COEFFICIENT – WIND SPEED

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The Sigma-0 histogram in Ku Band is reported in Figure 31. The S Band is no more being monitored due to the S Band Power drop, started on date 17<sup>th</sup> of January 2008, see section 7.2.1. The Sigma\_0 histogram in Ku Band is nominal. It shows secondary peaks as in the previous cycles. A small investigation on this problem, performed on the data of cycle 29, demonstrated that the backscattering distribution assumes a different behavior for different sea conditions [R-17]. Indeed the majority of the data is concentrated on lower values for rough sea state (southern hemisphere, winter conditions) and on higher values for calm sea state (northern hemisphere, summer conditions).



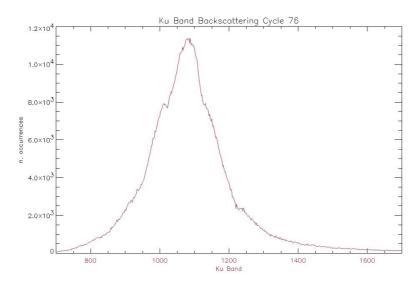


Figure 31: Histogram of Ku Band Backscattering Coefficient for cycle 76

In Figure 32, the backscattering coefficient daily average, computed for only ocean data, trend is reported. The Ku Band shows a nominal behavior.

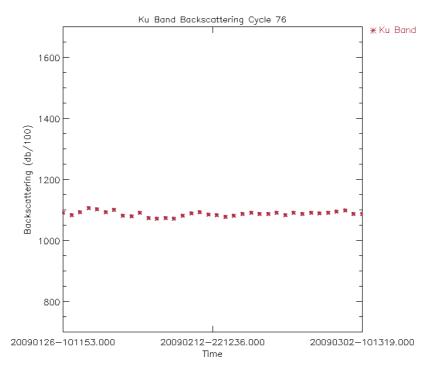


Figure 32: Ku Band Sigma\_0 daily average for cycle 76 (dB/100)

The histograms of Wind Speed computed for the Ku-band and the time behavior during the current cycle are reported in Figure 33 and 34, respectively. Given that the wind table has been updated



since IPF version 5.02, S.Abdallah Table is now used, the wind takes values between 1.18m/ and 21.30m/s.

The largest peak present in the histogram (about 50000 data for Wind < 1.2m/s) was removed from the plot in order to have the complete picture of the wind histogram.

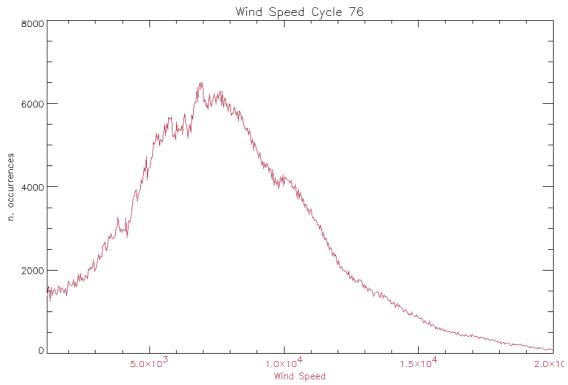


Figure 33: Histogram of Ku Wind Speed for cycle 76 (mm/sec)

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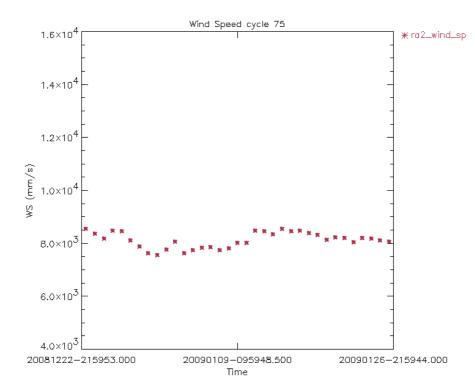


Figure 34: Ku Band Wind Speed daily average for cycle 76 (mm/s)

The Ku-Band Sigma\_0 trend, reported hereafter (Figure 35 and 35A), is characterized by a jump of on average 3.24 dBs concomitant with the operational up-load of IPF version 4.54 which occurred on the 9<sup>th</sup> of April 2003. This change is due to the upload of a new RA2\_CHD\_AX ADF file that artificially shifted the RA-2 real Sigma\_0 in order to align it with ERS-2 Sigma\_0 and make it coherent with the Witter and Chelton empirical wind model. A similar change in trend, but in the opposite direction, is also visible in the Wind Speed trend reported in Figure 36.

Beyond the huge jump that occurred in April 2003, the S-Band Sigma\_0 reports a smaller jump occurring on November the 26<sup>th</sup> 2003. Following the installation of the IPF processing chain V4.56, the average values of the RA-2 S-Band backscattering parameter, shows an increase of ~0.65 dBs, the new S-band sigma0 being higher with respect to the previous versions. The very low values of the S Band Backscattering around 30<sup>th</sup> July 2006 are related to the S Band Power Drop Anomaly occurred when the instrument was operating on RFFS B-side from 15 May until 21<sup>st</sup> June 2006 and from the 17<sup>th</sup> of January onwards due to the S Band Power Drop Anomaly affecting RFFS A-side. The S-Band Sigma\_0 daily means that are plotted outside the figure range can be traced back to the so-called S-Band anomaly (ref. par. 6.1.8).

In Figure 35A, the S Band is no more being monitored starting from cycle 65 due to the S Band Power drop, started on date 17<sup>th</sup> of January 2008, see section 7.2.1.



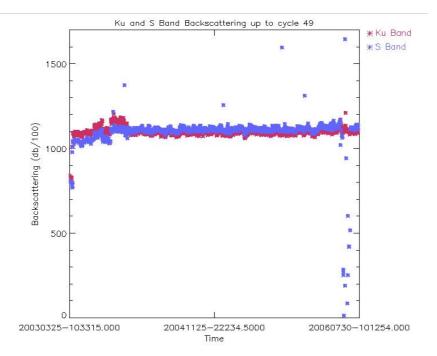


Figure 35: Ku and S band Backscattering daily averages up to cycle 49 (dB/100)

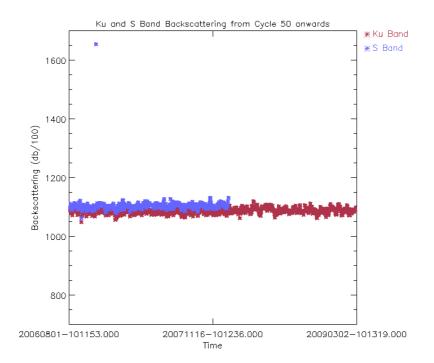


Figure 35A: Ku and S band Backscattering daily averages from cycle 50 onwards (dB/100)



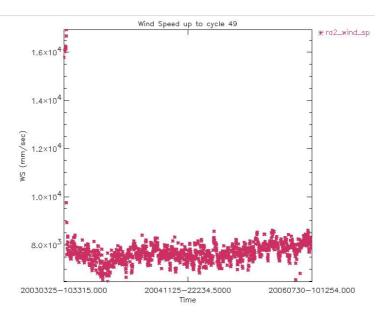


Figure 36: Wind Speed daily averages up to cycle 49 (mm/s)

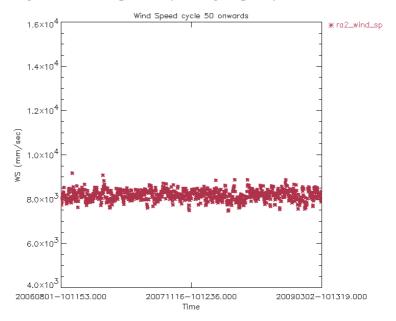


Figure 36A: Wind Speed daily averages from cycle 50 onwards (mm/s)

## 8 **PARTICULAR INVESTIGATIONS**

eesa

No particular investigations have been performed during the current cycle.



## **APPENDIX 1: IPF UPGRADES**

#### Table 4: L1B IPF version

IPF Version	Date of issue PDHSK& E, LRAC	L1B Algortihm upgrades	L1B ADF updates	ADF filename
V4.53	Nov. 27, 2002			
V4.54	Apr. 7, 2003	*Wrong sign in AGC calibration estimation *Missing integrity check for the Data Block number read from the Level 0 Data Blocks *The altitude above CoG and the altitude rate have to be included in the records also in case of dummy records *1Hz data should be referenced to data block 9.5 not block 10	Correction of the Tx-Rx gain of Ku and S band parameters (3.5 dB)	RA2_CHD_AX
V4.56	Nov. 26, 2003	1- Extrapolation of AGC value to the Waveform center (49.5) for both Ku- and Sband. 2 - Correction for an error found in the evaluation of S band AGC.	RA2 IF Mask	RA2_IFF_AX
V4.57	PDHS-K: 29-04-2004 PDHS-E: 28-04-2004			
V4.58	Aug. 9, 2004			
V5.0.2	Oct. 24, 2005	MWR Side Lobe correction upgrade	side lobe table and Config param	MWR_SLT_AX MWR_CON_AX
		USO clock period units correction	New ADF format - clock period unit	RA2_USO_AX RA2_CHD_AX RA2_CON_AX
		RA-2 alignment: OBDH & USO datation, IE flags correction		
		Rain Flag tunning to compensate for the increase of the S band Sigma0	New table in SOI file	RA2_SOI_AX
		Monthly IF estimation		RA2_IFF_AX
		Level 1B S-Band anomaly flag	New format	RA2_CON_AX
		DORIS Navigator CFI		



V5.03       Scp. 19, 2006       Level 1B S-Band anome to RAL Signal Content of the Window, delay (SPR-058).         V5.04       Jun. 20, 2007       DORIS Navigator the window spee signal to an error when applying the IP mask correction to the wate of the Window, delay (SPR-059).         V 5.06       Jun. 20, 2007       DORIS Navigator the work and to spee signal to an error when applying the IP mask correction to the Window, delay (SPR-059).         V 5.06       Jun. 20, 2007       DORIS Navigator threes of the work and to an error or the Window and to an error when applying the IP mask correction to the Window, delay (SPR-059).         V 5.06       Jun. 20, 2007       DORIS Navigator threes of the work and to spee signal to an error when applying the IP mask correction to the Wark Ar2/DORIS Alignment of Chain B to Prod Spee 3/N				
V 5.06Jun. 20, 2007DORIS Navigator threshold update to 900 seconds coverage RA2/DORIS Alignment of Chain B			MWR) Orbit Flag not well implemented: when a DORIS product is used for the processing, the Orbit flag is set to 1 for the whole length of the RA2 L1b product file while it should be set to 1 only for the part of the RA2 product overlapping with the DORIS one. Problem has been traced on OAR 1938 to be solved on next IPF delivery. Correction of the Rx_dist_fine from the Level 0 product, leading to an error in the calculation of the Window_delay (SPR-	
threshold update to 900 seconds coverage RA2/DORIS Alignment of Chain B	X17.00			
	V5.03	Sep. 19, 2006	anomaly flag well implemented Orbit Flag well implemented Correction of the Rx_dist_fine (for 80 and 20 MHz) from the Level 0 product, leading to an error when applying the IF mask correction on to the waveforms (SPR-	

#### Table 5: L2 IPF version

PF Version	Date of issue PDHS	L2 Algortihm upgrades	L2 ADF updates	ADF filename
V4.53	Nov. 27, 2002			
V4.54	Apr. 7, 2003			





V4.56	Nov. 26, 2003	SPR 26 Tuning of the Ice2 retracking New MWR NN algorithm	MSS CLS01 Rain flag Updated OCOG retracker thresholds Ice1/Sea Ice Conf file Sea State Bias Table file GOT00.2 Ocean Tide Sol 1 Map file FES 2002 Ocean Tide Sol 2 Map file FES 2002 Tidal Loading Coeff Map	RA2_MSS_AX RA2_SOI_AX RA2_ICT_AX RA2_SSB_AX RA2_OT1_AX RA2_OT1_AX RA2_OT2_AX RA2_TLD_AX
V4.57	PDHS-K: 29-04- 2004 PDHS-E: 28-04- 2004	ECMWF meteo files handling		
V4.58	Aug. 9, 2004	Addition of a Pass Number Field in FD Level		
	Oct. 24, 2005	<ul> <li>Handling of the new RA2_CHD_AX ADF</li> <li>Rain Flag tuning to compensate for the increase of the S band Sigma0</li> <li>Improving the mispointing estimation</li> <li>Export of the Level 1B S-band flag into the Level 2 data product</li> <li>Export of the Level 1B NRT orbit quality flag</li> <li>Addition of a Pass Number Field in FD Level 2 SPH product</li> <li>Addition of peakiness in Ku and S band in FDMAR</li> <li>Addition of square of the SWH in Ku and S band</li> <li>Correction of MCD flag</li> <li>SPH pass number (field 8) set to 0 in SPH NRT Level 2 data products</li> </ul>	New table in SOI file Two needed parameters in SOI file New format Addition of GOT2000.2 TLD New DEM AUX file (MACESS) merge of ACE land elevation data and Smith and Sandwell ocean bathymetry	RA2_CHD_AX RA2_SOI_AX RA2_SOI_AX RA2_SOI_AX RA2_SOI_AX
V 5.03	Sep. 19, 2006			
V 5.06	Jun. 20, 2007			





# **APPENDIX 2: AVAILABILITY:**

Start orbit	Stop orbit	Time [sec] instrum. Unavail- ability	Data Unav Time [sec]	Time [sec] L0 gaps	Time [sec] L1b gaps	Time [sec] L2 (FGD) gaps	% instrum. avail.	% data avail.	% L0 avail.	% L1b avail.	% L2 (FGD) avail.
36127	36227	0	1258	1258	1475	1487	100	99.99	99.79	99.75	99.75
36227	36327	0	24621	24621	24623	24632	100	99.99	95.92	95.92	95.92
36327	36427	0	4750	4750	4752	4760	100	99.99	99.21	99.21	99.21
36427	36527	0	1001	1001	1002	1014	100	99.99	99.83	99.83	99.83
36527	36627	0	1354	1354	1356	1369	100	99.99	99.79	99.79	99.79

### Table 6: RA-2 L0, L1b and L2 FGD Data products availability summary for Cycle 76

Table 7: MWR L0 Data products availability summary for Cycle 76

Start orbit	Stop orbit	Time [sec] instrum. unavailability	Time [sec] L0 gaps	% instrum. avail.	% L0 avail.
36127	36227	0	360	100	99.94
36227	36327	0	24168	100	96
36327	36427	0	4296	100	99.29
36427	36527	0	192	100	99.97
36527	36627	0	336	100	99.94

Table 8: DORIS L0 Data	products availability	summary for Cycle 76
------------------------	-----------------------	----------------------

Start orbit	Stop orbit	Time [sec] instrum. unavailability	Time [sec] L0 gaps	% instrum. avail.	% L0 avail.
36127	36227	0	31243	100	94.89
36227	36327	0	51641	100	91.46
36327	36427	0	35703	100	94.11
36427	36527	0	29965	100	95.05
36527	36627	0	31562	100	94.75



#### Table 9: List of gaps for RA-2 L0 Cycle 76

The list below only contains gaps higher then 200 seconds. Small gaps occurring everyday due to calibration or PDS anomalies have been suppressed.

RA2_ME0P							
start time	stop time	Duration (seconds)	Downlink				
08/02/2009 5.59	08/02/2009 6.05	328	36303@KS				
13/02/2009 17.52	13/02/2009 18.55	3770	36382@KS				
08/02/2009 4.26	08/02/2009 5.59	5577	36302@ES				
08/02/2009 2.50	08/02/2009 4.26	5794	36301@ES				
07/02/2009 23.53	08/02/2009 1.30	5810	36299@ES				
07/02/2009 22.11	07/02/2009 23.53	6129	36298@ES				

#### Table 10: List of gaps for MWR L0 Cycle 76

The list below only contains gaps higher then 200 seconds.

MWR_NL_0P							
start time	stop time	Duration (seconds)	downlink				
13/02/2009 17.48	13/02/2009 17.52	221	36381@KS				
08/02/2009 5.59	08/02/2009 6.04	314	36303@KS				
13/02/2009 17.52	13/02/2009 18.55	3763	36382@KS				
08/02/2009 2.50	08/02/2009 4.26	5794	36301@ES				
07/02/2009 23.53	08/02/2009 1.30	5797	36299@ES				
08/02/2009 4.26	08/02/2009 6.04	5891	36302@ES				
07/02/2009 22.11	07/02/2009 23.53	6129	36298@ES				

Table 11: List of gaps for RA-2 L1b Cycle 76 (gaps with respect to L0 products)The list of gaps below is referred to the L0 production and not to the planning.





RA2_MW1P						
predecessor name	start time	SIAN LIME II		ancestor coverage missing		
No gap						

## **APPENDIX 3: LEVEL 2 STATIC AUXILIARY DATA FILES**

AUX\_DEM\_AXVIEC20031201\_000000\_20031201\_000000\_20200101\_000000 AUX ATT AXVIEC20020924 131534 20020703 120000 20781231 235959 AUX\_LSM\_AXVIEC20020123\_141228\_20020101\_000000\_20200101\_000000 MWR LSF AXVIEC20020313 172218 20020101 000000 20200101 000000 MWR CHD AXVIEC20021111 131410 20020101 000000 20200101 000000 MWR\_LSF\_AXVIEC20020313\_172218\_20020101\_000000\_20200101\_000000 MWR SLT AXVIEC20050426 120000 20020101 000000 20200101 000000 RA2 IFA AXVIEC20050216 125529 20020101 000000 20200101 000000 RA2 IFB AXVIEC20050216 125738 20020101 000000 20200101 000000 RA2\_CHD\_AXVIEC20051017\_093900\_20020101\_000000\_20200101\_000000 RA2\_CST\_AXVIEC20020621\_135858\_20020101\_000000\_20200101\_000000 RA2 DIP AXVIEC20020122 134206 20020101 000000 20200101 000000 RA2 GEO\_AXVIEC20020314\_093428\_20020101\_000000\_20200101\_000000 RA2\_ICT\_AXVIEC20031208\_143628\_20020101\_000000\_20200101\_000000 RA2 IOC AXVIEC20020122 141121 20020101 000000 20200101 000000 RA2 MET AXVIEC20020204 073357 20020101 000000 20200101 000000 RA2 MSS AXVIEC20031208 145545 20020101 000000 20200101 000000 RA2\_OT1\_AXVIEC20040120\_082051\_20020101\_000000\_20200101\_000000 RA2\_OT2\_AXVIEC20031208\_150159\_20020101\_000000\_20200101\_000000 RA2 SET AXVIEC20020122 150917 20020101 000000 20200101 000000 RA2\_SL1\_AXVIEC20030131\_100228\_20020101\_000000\_20200101\_000000 RA2\_SL2\_AXVIEC20030131\_101757\_20020101\_000000\_20200101\_000000 RA2\_SOI\_AXVIEC20051003\_170000\_20020101\_000000\_20200101\_000000 RA2 SSB AXVIEC20051129 111810 20020101 000000 20200101 000000 RA2\_TLD\_AXVIEC20031208\_151137\_20020101\_000000\_20200101\_000000 RA2\_TLG\_AXVIEC20040310\_110000\_20020101\_000000\_20200101\_000000





# **APPENDIX 4: SIGMA0 ABSOLUTE CALIBRATION**

 Table 18: Transponder measurement results up to cycle 76

Absolute Orbit nb	Date of Measurement	Location / Rel. track	RA-2 resolution	Transponder Bias [dB]	ECMWF Wet Tropo. Corr. [dB]
10389	24-feb-04	Rome / 315	Low	1,552	0,120
10511	04-mar-04	Valmontone / 437	Low	1,542	0,102
10618	11-mar-04	Fiuggi / 43	Low	1,447	0,135
10783	23-mar-04	Maccarese / 208	Low	1,540	0,142
10890	30-mar-04	Rome / 315	Low	1,442	0,152
11119	15-apr-04	Fiuggi / 43	High	0,963	0,122
11513	13-mag-04	Valmontone / 437	Low	1,353	0,133
11620	20-mag-04	Fiuggi / 43	Low	1,427	0,139
11892	08-giu-04	Rome / 315	Low	1,504	0,154
12014	17-giu-04	Valmontone / 437	Low	1,448	0,348
12121	24-giu-04	Fiuggi / 43	Low	1,576	0,149
14290	23-nov-04	Maccarese / 208	Low	1,43	0,164
14397	30-nov-04	Rome / 315	Low	1,11	0,142
14519	9-dic-04	Valmontone / 437	Low	1,26	0,248
14791	28-dic-04	Maccarese / 208	High	0,97	0,134
14898	4-gen-05	Rome / 315	High	0,95	0,114
15020	13-gen-05	Valmontone / 437	High	0,88	0,118
15127	20-gen-05	Fiuggi / 43	High	1,01	0,108
15292	1-feb-05	Maccarese / 208	High	0,95	0,132
15399	8-feb-05	Rome / 315	High	1,05	0,124
15521	17-feb-05	Valmontone / 437	High	0,94	0,115
15793	8-mar-05	Maccarese / 208	High	0,93	0,116
15900	15-mar-05	Rome / 315	High	0,93	0,128
16022	24-mar-05	Valmontone / 437	High	0,94	0,154
16294	12-apr-05	Maccarese / 208	High	0,97	0,140
16401	19-apr-05	Rome / 315	High	0,99	0,134
16523	28-apr-05	Valmontone / 437	High	0,97	0,114
16795	17-may-05	Maccarese / 208	High	0,84	0,168
16902	24-may-05	Rome / 315	High	1,00	0,152
17403	28-jun-05	Rome / 315	High	1,13	0,16
17525	7-jul-05	Valmontone / 437	High	1,04	0,13
17904	02-aug-05	Rome / 315	High	1,02	0,188
18026	11-aug-05	Valmontone / 437	High	0,93	0,154
18405	06-sep-05	Rome / 315	High	1,06	0,16
18634	22-Sep-05	Fiuggi/43	High	1,00	0,152
18799	04-Oct-05	Maccarese/208	High	0,85	0,164
18906	11-Oct-05	Perm site Rome / 315	Low	1,46	0,156
19407	15-Nov-05	Perm site Rome / 315	High	1,09	0,19
20409	24-Jan-06	Perm site Rome / 315	High	1,38	0,110
20910	28-Feb-06	Perm site Rome / 315	High	0,98	0,124
21912	9-May-06	Perm site Rome / 315	High	1,0	0,138
23916	26-Sep-06	Perm site Rome / 315	High	1,05	0,172



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24417	31-Oct-06	Perm site Rome / 315	High	1,08	0,146
24918	05-Dec-06	Perm site Rome / 315	High	1,00	0,156
25419	09-Jan-2007	Perm site Rome / 315	High	0,97	0,148
25929	13-Feb-2007	Perm site Rome / 315	High	1,07	0,118
26922	24-Apr-2007	Perm site Rome / 315	High	1,17	0,154
27423	29-May-2007	Perm site Rome / 315	High	1,04	0,168
29928	20-Nov-2007	Perm site Rome / 315	High	1,04	0,139
30930	29-Jan-2008	Perm site Rome / 315	High	1,01	0,013
31431	04-Mar-2008	Perm site Rome / 315	High	1,04	0,139
32433	13-May-2008	Perm site Rome / 315	High	0,87	0,1758
33435	22-Jul-2008	Perm site Rome / 315	High	1,0	0,1356
35439	09-Dec-2008	Perm site Rome / 315	High	0,98	n/a
36441	17-Feb-2009	Perm site Rome / 315	High	0,97	0,0803

### **APPENDIX 5: S-BAND ANOMALY**

RA-2 S Band has failed on cycle 65.

## **APPENDIX 6: IE SITES COORDINATES**

ZONE ID="CapraiaA"
ZONE_ID= 'CapraiaA'           RECORD polygon_pt: LONG=+009.934000 <deg> LAT=+042.970000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+009.863000 <deg> LAT=+042.970000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+009.863000 <deg> LAT=+043.166000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+009.934000 <deg> LAT=+043.166000<deg></deg></deg>
ENDRECORD
ZONE_ID="Toulon_D"
RECORD polygon_pt: LONG=+005.500000 <deg> LAT=+043.070000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+005.473000 <deg> LAT=+043.070000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+005.473000 <deg> LAT=+043.160000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+005.500000 <deg> LAT=+043.160000<deg></deg></deg>
ENDRECORD
ZONE_ID="Vostok_x"
RECORD polygon_pt: LONG=+106.500000 <deg> LAT=-078.000000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+105.500000 <deg> LAT=-078.000000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+105.500000 <deg> LAT=-077.500000<deg></deg></deg>
ENDRECORD
RECORD polygon_pt: LONG=+106.500000 <deg> LAT=-077.500000<deg></deg></deg>



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ENDRECORD
-----------

ZONE\_ID="Dome\_x\_\_"

RECORD polygon\_pt: LONG=+124.000000<deg> LAT=-075.250000<deg>

ENDRECORD

RECORD polygon\_pt: LONG=+122.000000<deg> LAT=-075.250000<deg>

ENDRECORD

RECORD polygon\_pt: LONG=+122.000000<deg> LAT=-074.750000<deg>

ENDRECORD

RECORD polygon\_pt: LONG=+124.000000<deg> LAT=-074.750000<deg> ENDRECORD